## **UNCLASSIFIED**

# AD NUMBER AD875818 LIMITATION CHANGES TO: Approved for public release; distribution is unlimited. FROM: Distribution authorized to U.S. Gov't. agencies only; Administrative/Operational Use; OCT 1970. Other requests shall be referred to Office of Naval Research, Code 462, Washington, DC 20360. AUTHORITY onr ltr, 15 mar 1979

THIS REPORT HAS BEEN DELIMITED AND CLEARED FOR PUBLIC RELEASE UNDER DOD DIRECTIVE 5200.20 AND NO RESTRICTIONS ARE IMPOSED UPON ITS USE AND DISCLOSURE.

DISTRIBUTION STATEMENT A

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED.



ADL-72580-2

TECHNICAL REPORT 2

TIME CORRECTION IN PASSIVE RANGING: BREAKTHROUGH OR BOOTSTRAP?

James M. Dobbie

Under Contract

N00014-70-C-0322

NR 364-025/2-12-70 (462)

Prepared For

Office of Naval Research Department of the Navy Washington, D. C. 20360

Prepared By

Arthur D. Little, Inc.
Acorn Park
Cambridge, Massachusetts 02140

October 1970

"REPRODUCTION IN WHOLE OR IN PART IS PERMITTED FOR ANY PURPOSE OF THE UNITED STATES GOVERNMENT."



# TIME CORRECTION IN PASSIVE RANGING: BREAKTHROUGH OR BOOTSTRAP?\*

#### Introduction and Summary

The concept of time correction in passive ranging is described in reference (a). It was developed by Daniel H. Wagner, Associates, who believe it to be a breakthrough in the subject of passive ranging. The concept has been applied to the four-bearings problem for which it was conceived and to the Ekelund range estimate involving two bearing rates. A time-corrected Ekelund range estimator is being considered for part of the target motion analysis in a new fire-control system.

Does the concept of time correction represent a significant breakthrough, as claimed, or is the apparent gain an illusion? We have examined this question carefully and have found no support for the claims that have been made for the method. We have concluded that the method has no special value in passive ranging and should be discarded. Our arguments are presented below.

#### The Time-Correction Method

Time correction, as developed in reference (a), is a method for the computation of a range R\* at a time t\* from the four bearings  $B_1$ ,  $B_2$   $B_3$ ,  $B_4$  observed at the corresponding times  $t_1$ ,  $t_2$ ,  $t_3$ ,  $t_4$  which are assumed to be in the order  $t_1 < t_2 < t_3 < t_4$ . We outline the derivation of the equations below.

This research was supported by the Office of Naval Research under Contract No. N00014-70-C-0322.

We use the standard naval coordinate system. For i = 1, 2, 3, 4, let

t, = time at the ith observation

B, - bearing of the target from the tracking submarine

R, = range of the target from the tracking submarine

uri = range component of target velocity measured outward along B,

ubi = normal component of target velocity measured normal to B in the direction of positive (clockwise) angular rotation

Also, let

 $\mu_{ij}^{k}$  = distance (directed) tracking submarine moves in the direction  $B_{k}$  from time  $t_{i}$  to time  $t_{j}$ 

 $v_{ij}^{k}$  = distance (directed) tracking submarine moves normal to  $B_{k}$  from time  $t_{i}$  to time  $t_{j}$ 

It will be convenient to use

$$t_{ij} = t_j - t_i$$
,  $B_{ij} = B_j - B_i$ ,  $S_{ij} = \sin B_{ij}$ ,  $C_{ij} = \cos B_{ij}$ 

Assuming linear target motion it is easy to derive the equations

$$R_2 S_{12} = u_{b1} t_{12} - v_{12}^1$$
 (1)

$$R_4 S_{34} = u_{b3} t_{34} - v_{34}^3$$
 (2)

$$R_2 C_{24} = R_4 - u_{r4} t_{24} + \mu_{24}^4$$
 (3)

$$u_{b1} c_{13} = u_{b3} + u_{r1} s_{13}$$
 (4)

These equations are equivalent to the equations (2-5) (2-6) (2-7) and an unnumbered equation on page 2-18 of reference (a). Eliminating  $R_2$ ,  $u_{11}$ , and  $u_{12}$  from the above equations and solving

Eliminating  $\mathbf{R}_2,~\mathbf{u}_{b1},~\text{and}~\mathbf{u}_{b3}$  from the above equations and solving for  $\mathbf{R}_4$  we obtain

$$R_4 = \hat{R}_4 + a_1 u_{r1} + a_2 u_{r4}$$
 (5)

where

$$\hat{R}_{4} = (v_{34}^{3} / t_{34} - c_{13} v_{12}^{1} / t_{12} - \mu_{24}^{4} \beta_{12}) / (\beta_{12} - \beta_{34})$$
 (6)

$$a_1 = s_{13} / (\beta_{12} - \beta_{34}), a_2 = t_{24} \beta_{12} / (\beta_{12} - \beta_{34})$$
 (7)

$$\beta_{12} = (S_{12}/t_{12}) (C_{13}/C_{24}), \beta_{34} = S_{34}/t_{34}$$
 (8)

Equation (5) expresses the range  $\mathbf{R}_4$  in terms of an estimate  $\hat{\mathbf{R}}_4$  that can be computed from observed quantities, and two range components of target velocity.

In the time-correction method the two range components,  $u_{r1}$  and  $u_{r4}$ , of target velocity are eliminated by introducing two bearings,  $B_5$  and  $B_6$ , at times  $t_5$  and  $t_6$  by using two equations of type (3) above. They are

$$R_5 C_{45} = R_4 + u_{r4} t_{45} - \mu_{45}^4$$
 (9)

$$R_{6} C_{16} = R_{5} C_{15} + u_{r1} t_{56} - \mu_{56}^{1}$$
 (10)

The general equation of this type is

$$R_{j} C_{jk} = R_{i} C_{ik} + u_{rk} t_{ij} - \mu_{ij}^{k}$$
(11)

Equations (3) (9) (10) are special cases of equation (11).

Eliminating  $\mathbf{R}_4$  and  $\mathbf{R}_5$  from equations (5) (9) (10) we obtain the equation

$$R_6 C_{16} = (C_{15}/C_{45}) (\hat{R}_4 - \mu_{45}^4) - \mu_{56}^1 + a_3 u_{r1} + a_4 u_{r4}$$
 (12)

where  $a_3$  and  $a_4$  depend on  $t_5$  and  $t_6$ . Let  $\overline{t}$  and  $t^*$  be the values of  $t_5$  and  $t_6$  respectively that make  $a_3 = a_4 = 0$ . Also, let  $\overline{B}$  and  $\overline{B}$  be the corresponding bearings of the target at these times, and let  $\overline{R}$  be the corresponding value of  $\overline{R}_6$ . Then

$$\overline{t} = (\beta_{12}t_2 - \beta_{34}t_4) / (\beta_{12} - \beta_{34})$$
 (13)

$$t^* = \overline{t} - S_{13} \cos (\overline{B} - B_1) / (\beta_{12} - \beta_{34}) \cos (\overline{B} - B_4)$$
 (14)

$$R^* \cos (B^* - B_1) = (\hat{R}_4 - \mu_{45}^4) \cos (\overline{B} - B_1) / \cos (\overline{B} - B_4) - \mu_{56}^1$$
(15)

Equations (13) (14) (15) are the same as equations (2-14) (2-18) (2-20) respectively of reference (a) when our subscript sequence (1, 2, 3, 4) is replaced by the sequence (1', 1, 2', 2) of reference (a). (It is convenient to retain the subscripts 5 and 6 in  $\mu_{45}^4$  and  $\mu_{56}^1$  to avoid ambiguity.)

The range R\* is called "the time-corrected range" and t\* is called "the best range time." The time-correction method consists of using equation (15) to compute R\* and equation (14) to compute the corresponding time t\* at which it applies.

#### Discussion

Does the above method offer a significant breakthrough in passive ranging? Various claims are made for the method in reference (a). It is asserted that the basis for the method is that "every ranging maneuver has associated with it a best range time", at which time the error in the range component of velocity "will have minimum effect on the accuracy of the range solution." And when t is in the future, it is claimed that the four bearing observations and the three distance measurements ( $\mu_{24}^4$ ,  $\mu_{45}^4$ , and  $\mu_{56}^1$ ) "provide an accurate range at a future time with no additional information needed." Are these claims justified?

The derivation is deceptively simple. No optimization is needed. Only simple algebraic manipulations of well-known equations are required to eliminate the two range components,  $u_{r1}$  and  $u_{r4}$ , of target velocity from the range equation (5). It is surprising that such a simple derivation yields a range estimate that is

superior to other range estimates, as claimed. And it is even more surprising that the result remained undiscovered so long, since the derivation involves only simple mathematics.

Since t is not obtained by optimizing a payoff function, why is it called "best"? Apparently, t is called the best range time because the corresponding range R has been written in equation (15) in a form that does not involve a range component u of target velocity. Immediately following the equation, which is numbered (2-20) in reference (a), it is stated that, "The range R is the time-corrected range. Note that all of the terms in Equation (2-20) can be measured by the SSK. The time t at which this applies is called the best range time..."

Can all of the terms in equation (15) above be "measured"? Certainly,  $\overline{B}$  and  $\overline{B}$  can't be measured in the same way that  $\overline{B}_1$ ,  $\overline{B}_2$ ,  $\overline{B}_3$ ,  $\overline{B}_4$  are measured by recording the observed value at the time of occurrence. The bearings  $\overline{B}$  and  $\overline{B}$  are the bearings at the times  $\overline{t}$  and  $\overline{t}$ . Obviously, if  $\overline{t}$  and  $\overline{t}$  fall outside the tracking interval  $(t_1, t_4)$ , the corresponding bearings  $\overline{B}$  and  $\overline{B}$  can't be observed. Nor can they be observed, if  $\overline{t}$  and  $\overline{t}$  fall inside the tracking interval. The values of  $\overline{t}$  and  $\overline{t}$  in equations (13) and (14) require the values of all four bearings. Hence,  $t_4$  is the earliest time at which  $\overline{t}$  and  $\overline{t}$  can be computed, and it then is too late to observe  $\overline{B}$  and  $\overline{B}$ .

Hence,  $\overline{B}$  and  $\overline{B}^*$  must be "estimated" from the bearing-time plot, by interpolation when  $\overline{t}$  and  $\overline{t}^*$  fall inside the tracking interval and by extrapolation when they fall outside the tracking interval. In either case the errors in estimating  $\overline{B}$  and  $\overline{B}^*$  may be large. Here, interpolation may be as inaccurate as extrapolation, since the SSK is maneuvering during the tracking interval and bearing observations usually are not made during the turns. However, the resulting errors in the cosine factors, and in  $\overline{R}^*$ , in equation (15) would be small. Hence, the fact that  $\overline{B}$  and  $\overline{B}^*$ 

must be estimated by interpolation or extrapolation is not a major disadvantage. The error is comparable to that made in using the small-angle approximations.

Errors also will occur in estimating the two distances  $\mu_{45}^4$  and  $\mu_{56}^1$ , since they also must be obtained by interpolation or extrapolation after  $\bar{t}$  (= $t_5$ ) and  $t^*$  (= $t_6$ ) have been computed. These errors may produce larger errors in  $R^*$  than those produced by the errors in estimating  $\bar{B}$  and  $\bar{B}^*$ .

More serious questions concerning the method can be raised as follows:

- a. What is the motivation behind the procedure? What objective are we trying to achieve? Is it merely to find an equation for the range R\* at some time t\* in a form that involves only quantities that the SSK can measure or estimate?
- b. Does the procedure lead to a unique solution that has some desirable property? If so, what?
- c. After we find R\* at time t\*, how will we obtain an estimate of the range R at time t when the time t at which we need a range estimate does not coincide with t\*? Will this require that we use an equation of type (11) above that requires a range component u<sub>r</sub> of target velocity? If so, what have we gained by first computing R\*?

If the objective of the method is to obtain an equation for range in a form that involves only quantities that the SSK can measure or estimate, why stop at one? There are many such equations. What special properties does the "solution" in equation (15) have that it should be selected?

Other "solutions" of the type displayed above can be obtained by interchanging two subscripts. This possibility seems to have been recognized in some parts of reference (a), and used in a special case in chapter 4. But the implications concerning the validity of the general method and the uniqueness of the solution were not explored.

Some of the "solutions" obtained in this way reduce to the same forms for  $t^*$  and  $R^*$  in the small-angle approximations, and others do not. For example, if we replace equations (9) and (10) by the equations

$$R_5 C_{15} = R_4 C_{14} + u_{r1} t_{45} - \mu_{45}^1$$
  
 $R_6 C_{46} = R_5 C_{45} + u_{r4} t_{56} - \mu_{56}^4$ 

and repeat the steps of the solution, the equations for  $t^*$  and  $R^*$  are not identical to equations (14) and (15) but reduce to the same small-angle forms that equations (14) and (15) become. The equation for t does not do so. On the other hand, if we interchange the subscripts 2 and 3 in the original solution, we get a solution for which the small-angle approximations for  $t^*$  and  $R^*$ , as well as that for t, do not reduce to the same small-angle forms obtained in the original solution. This solution is written in full below:

$$\overline{t} = (\beta_{13} t_3 - \beta_{24} t_4) / (\beta_{13} - \beta_{24})$$
 (13a)

$$t^* = \overline{t} - S_{12} \cos (\overline{B} - B_1) / (\beta_{13} - \beta_{24}) \cos (\overline{B} - B_4)$$
 (14a)

$$R^* \cos (B^* - B_1) = (\hat{R}_4 - \mu_{45}^4) \cos (\overline{B} - B_1) / \cos (\overline{B} - B_4) - \mu_{56}^1$$
(15a)

where now

$$\hat{R}_{4} = (v_{24}^{2} / t_{24} - C_{12} v_{13}^{1} / t_{13} - \mu_{34}^{4} \beta_{13}) / (\beta_{13} - \beta_{24})$$
 (6a)

$$\beta_{13} = (S_{13} / t_{13}) (C_{12} / C_{34}), \beta_{24} = S_{24} / t_{24}$$
 (8a)

It is evident that we can't call the time t in equation (14) the best range time, since it is neither unique nor optimal in any demonstrated way. It may be a good range time, but the property, if any, that makes it better than an arbitrary time has not been displayed in reference (a) or elsewhere, to our knowledge. The property that led to its introduction - that it permits us to write an equation for the corresponding range R in a form that involves only quantities that the SSK can measure or estimate - is not a sufficient reason for its use. Any arbitrary time has this property.

Under the assumption of linear target motion, on which the derivations above and in reference (a) are based, four bearing observations (made on a ranging maneuver that doesn't consist of a single linear path) are sufficient for a complete TMA. Hence, the range R at any <u>arbitrary</u> time t can be written as a function of t, the four observation pairs  $(t_1, B_1)$ , i = 1, 2, 3, 4, and components of own-ship motion. It is not necessary to compute two times  $\bar{t}$  and  $\bar{t}$  and then estimate the corresponding bearings  $\bar{B}$  and  $\bar{B}$  by interpolation or extrapolation from the bearing-time plot. The only bearings involved are  $B_1$ ,  $B_2$ ,  $B_3$ ,  $B_4$ .

The TMA requires only the solution of four simultaneous linear equations in a set of target parameters, such as  $x_4$ ,  $y_4$ ,  $u_x$ ,  $u_y$ , where  $u_x$  and  $u_y$  are the components of target velocity. The solution has been obtained many times. In fact, Spiess' form of the solution is discussed in chapter 5 of reference (a). By means of the TMA we can write an equation for the range R at an arbitrary time t, in a form that involves only quantities that the SSK can measure or estimate.

The solution for R is simple when t coincides with one of the observation times. For example, when  $t = t_4$ , the solution becomes

$$R_{4} = \frac{v_{14}^{1} t_{24} t_{34} s_{23} - v_{24}^{2} t_{14} t_{34} s_{13} + v_{34}^{3} t_{14} t_{24} s_{12}}{-t_{13} t_{24} s_{12} s_{34} + t_{12} t_{34} s_{13} s_{24}}$$
(16)

where, as before,

We do not recommend the use of the estimator (16) for  $R_4$ . However, equation (16) is a simple equation for the range  $R_4$  in terms of the four  $(t_1, B_1)$  pairs and three components of own-ship motion. Also, it is exact, that is, it will yield the exact value for the range at time  $t_4$  in the absence of errors in the measured quantities. And, certainly, the SSK can measure all the quantities in equation (16). Do these facts make  $t_4$  a "good" range time? If not, what property does  $t^*$  possess that makes it better? Will any gain that is obtained from using  $t^*$  and  $t^*$  be lost in computing  $t_4$  from  $t^*$ , if  $t_4$  is the time at which a range estimate is needed.

It is surprising that nothing is said in reference (a) about the problem of converting from the estimate  $R^*$  at time  $t^*$  to an estimate of the range at the time at which it is needed. How is this conversion to be made? By solving for the TMA? If so, won't we finish with the same deterministic relationships, but in a more circuitous form, with many more chances for errors in the measurements? Will the estimate of  $R_4$  by such a method be more accurate than that obtained from equation (16)?

A possible way of avoiding these questions is to state that the submarine commander can control the value of  $t^*$  by his choice of ranging maneuvers and hence can make  $t^*$  equal to, or close to, the desired time, say  $t_4$ . In fact, this point is offered in

reference (a) as a major advantage of the time-correction method. We contend that control of the value of t is not a practical solution to this problem. The commander of own submarine has only partial control over t and that control is rather tenuous. The commander of the target submarine must cooperate by maintaining constant course and speed. Even when he does so, the problem of the selection of maneuvers and times t<sub>1</sub>, t<sub>2</sub>, t<sub>3</sub>, t<sub>4</sub> that will make t fall at t<sub>4</sub> is not an easy one to solve, since the value of t can't be computed until the last bearing B<sub>4</sub> has been recorded. It is the type of problem that can be solved after the fact but is very difficult to solve in real time and sequence. Even if it can be done, there remains the problem of how to assure linear target motion in the relevant time interval.

#### Criteria For Selection of a Range Estimator

The equations derived above have been based on two important assumptions, as follows:

- (1) The target submarine maintains constant course and speed throughout the relevant time interval, which is the maximum interval spanned by the times involved in the computation.
- (2) There are no errors in the measurements of bearings, including those obtained by interpolation or extrapolation, or in the measurement of own-submarine motion.

Both assumptions are questionable, at best, and we should consider the effects of departures from these assumptions.

The argument so far has been based solely on formula accuracy under these assumptions. The main argument for the use of  $R^*$  in equation (15) seems to be that equation (15) is exact and contains only quantities that the SSK can measure or estimate;

whereas some other equations for a range estimator are inexact, such as equation (6) as an estimator for  $R_4$ ; or they contain quantities that the SSK can't measure or estimate, such as equation (4). More precisely, equation (4) contains two factors,  $u_{r1}$  and  $u_{r4}$ , that the analyst hasn't taken the trouble to express in terms of quantities that the SSK can measure or estimate. If this had been done, equation (5) would have reduced to equation (16) when simplified.

Formula accuracy is only one characteristic of a range estimator, and not the dominant one in most applications. The main characteristics of a range estimator are:

- (a) Formula accuracy
- (b) Ruggedness to target maneuvers
- (c) Sensitivity to error measurements

It is not likely that an estimator will dominate all other contenders in all respects. For example, the estimator for  $R_4$  in equation (16) is exact, and hence superior in formula accuracy to the estimator  $\hat{R}_4$  for  $R_4$  in equation (6), and to the estimator  $R_E$  obtained by dropping the  $\mu_{24}^4$  term in equation (6), and to the Ekelund estimator  $R_{EK}$  (a limiting form of  $R_E$ ), since  $\hat{R}_4$ ,  $R_E$ , and  $R_{EK}$  are biased estimators. However,  $R_4$  in (16) probably is not as rugged to target maneuvers as some of the other estimators, and undoubtedly is more sensitive to random errors than  $\hat{R}_4$ ,  $R_E$ , or  $R_{EK}$ . It is impossible to estimate the ruggedness and sensitivity, or even the formula accuracy, of an estimator for  $R_4$  that starts with  $R^*$  in equation (15), since it depends on the procedure used to calculate  $R_4$  from  $R^*$ .

If there is no dominant estimator, how do we select one? We believe that this decision requires a careful examination of the use that will be made of the range estimate and the selection of a suitable performance measure or payoff function. We then

compute the payoff function for possible target maneuvers and error distributions, select the dominant estimator, if it exists, or use a weighted mean otherwise.

In many applications formula accuracy is relatively unimportant, particularly when likely target maneuvers or measurement errors will produce range errors that are much larger than the inherent bias in the formula. Under such conditions it is absurd to select an estimator on the criterion of formula accuracy, alone or predominantly.

Comparisons of range estimators, or complete TMA estimators, often (usually?) have been made on a partial basis. We believe that all characteristics of the estimator should be considered. It should be possible to analyze and compare leading contenders in respect to the major characteristics listed above. When this has been done, and done impartially and completely, we will have available for the first time the facts that are necessary for the intelligent selection of an estimator. And the selection should be made in consideration of the use that will be made of the estimate and the characteristics of the estimator that are important for that application.

In many applications of passive tracking the entire TMA is needed. For this reason we believe that the analysis should be extended to TMA estimators, as well as range estimators.

A complete analysis of the type described above seldom has been attempted, and perhaps never has been done satisfactorily. An analysis of several range estimators (the Ekelund and a new three bearing-rate estimator that removes the bias in the Ekelund estimator) was made in reference (b). Reference (b) also includes analyses of several TMA estimators (CHURN and a new one developed by the writer), using acquisition probability as the payoff function. While these analyses cover all the major characteristics

of the estimators, they are not extensive enough to be considered complete.

### References:

- (a) COMSUBDEVGROUP TWO Report No. 2-69, Passive Ranging Manual (U), Volume III THEORY, Dec. 1969 Confidential.
- (b) ADL Report NUWRES #12, "Control Modes and Acquisition Probabilities for Torpedo MK 48 (U), "Contract No. N00140-68-C-0278, January 1970, Final Report Confidential, Technical Appendices (bound separately) Unclassified.

	ROL DATA - R & D					
(Security classification of title, body of abstract and indexing						
1. ORIGINATING ACTIVITY (Corporate author) Arthur D. Little, Inc.	20.	28. REPORT SECURITY CLASSIFICATION				
Acorn Park	26.	UNCLASSIFIED				
Cambridge, Massachusetts 02140						
3. REPORT TITLE						
Time Correction in Passive Ranging: Break	through or Boot	strap?				
4. DESCRIPTIVE NOTES (Type of repet) and inclusive dates)						
Technical Report 2	·					
6. AUTHOR(S) (First name, middle initial, last name)						
James M. Dobbie						
October 1970	78. TOTAL NO. OF PA	GES	76. NO. OF REFS			
M. CONTRACT OF GRANT NO.	SE ORIGINATOR'S RE	PORT NUME	I DER(8)			
N00014-70-C-0322	ADL-72580-2					
A. PROJECT NO.						
c.	9b. OTHER REPORT NO(8) (Any other numbers that may be assigned					
	this report)					
d.	Technical Report 2					
10. DISTRIBUTION STATEMENT	the econolog of	f the II	S Covernment			
Each transmittal of this document outside must have prior approval of the Office of	Naval Research	Code 46	2 or the Naval			
Underwater Systems Center/Newport.						
11. SUPPLEMENTARY NOTES	12. SPONSORING MILITARY ACTIVITY					
	Naval Analysis Programs Office of Naval Research					
	Dept. of Navy, Washington D.C. 20360					
13. ABSTRACT						
A review is made of the method of time cor	rrection in pass	sive ran	iging. It is			
concluded that there is little, if any, su	ipport for the	claims m	ade for the			
method.						

DD FORM 1473 REPLACES DO FORM 1475, 1 JAN 64, WHICH IS

UNCLASSIFIED
Security Classification

UNCLASSIFIED
Security Classification

14. KEY WORDS	LIN	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT	
Time Correction							
Passive Ranging			l	}			
			1				
	1						
	1						
	1						
	1						
			·				
l l							
	i				1		
	l li						
8	1				1		
		n 10					

UNCLASSIFIED
Security Classification